EXPERIMENTAL AND NUMERICAL STUDY OF THE CLOUD CAVITATING FLOW AROUND A SLENDER CYLINDER WITH A PETALS-SHAPED SECTION

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ABSTRACT

Three-dimensional flow structures of sheet-to-cloud transition have attracted a lot of attentions in the past years. Cloud cavitating flow around a slender cylinder with the petals shaped section was investigated by experimental and numerical methods. The experiment was performed in a launching system on the basis of the SHPB technology with a high-speed camera. The numerical method was established with large-eddy simulation (LES) and volume of fraction (VOF) approach. Numerical results agree well with experimental data. Interesting phenomena of three-dimensional evolution are observed, especially in which the U-type shedding of cavities are generated repeatedly in the channels along the inner region. Analysis of the unsteady evolution of cavitating flow is performed, and emphasis is placed on the study of the relationship between flow structures around the ridges and channels. Results demonstrate that the pressure gradient variation in the tangential direction of the cylinder is essential for the unsteady development of the cavity including re-entry jet and local shedding. Interactions between the cavity and vortex are also analyzed, which indicates that the U-type cavity/vortex shedding is jointly induced by the re-entry jet and tangential vortex.

INTRODUCTION

Cavitation instability has been always widely concerned because of the negative impacts possibly caused on hydraulic machinery such as vibrations and erosions[1, 2]. Many researchers have investigated the fundamental physics of cavitation phenomenon concerning the evolution of both sheet and cloud cavitation on simple objects with two-dimensional and axisymmetric geometries such as hydrofoils, wedges and cylinders[3-12]. Typical quasi-periodic phenomena including growth, re-entry jet, shedding, collapse is in-depth investigated. Especially, Re-entry jet has been revealed as a key factor for
sheet to cloud cavitation transition of cavities by experimental and numerical methods in recent studies.

In engineering applications, the cavitating flow often occurs around objects with more complex shapes, and three-dimensional flow structures with multi-scale features are involved. For example, unsteady cavitating flows around twisted hydrofoils were widely investigated [13-15], which were constructed by using sections with variable attack angle in the span wise direction. Multi-scale U-type cavity vortex structures were observed in experimental and numerical results, and the secondary cavity shedding with small scale is demonstrated to have an important impact on the primary cavity shedding. Non-axisymmetric disks like petals/star shapes were used to investigate the perturbations on collapse and pinch-off characteristics of air cavity in water [16]. There are also many researches which were carried out on issues with complex cavity structures such as highly skewed propellers [17-19], turbomachinery [20, 21], and so on.

In the present paper, cloud cavitation flow around a slender cylinder with petals shaped section was investigated by experimental and numerical methods. Unsteady evolutions of the cavity and vortex structures are obtained and analyzed. Special emphasis is placed on discussing the interaction between flows in the inner channels and outer ridges, as well as the shedding cavity vortex structures induced by their interaction.

1. EXPERIMENTAL SETUP

The experiment was performed in a launching system on the basis of the Split Hopkinson Pressure Bar (SHPB) technology with a high-speed camera [22]. In this study, the test model is a slender cylinder with a petal-shaped section (as shown in Fig.1). This shape is chosen as a typical disturbance form of the circle cylinder [16]. The outer diameter is 37 mm and the inner diameter is 60% of the outer diameter. There are six ridges and channels arranged around the cylinder axis uniformly. The overall length is 200 mm, while the rear half of the model is a circular cylinder. The model is initially put on a support in the middle of the test section. Its material is polished stainless steel.

The experiment system is shown in Fig.2, and the details of the experiment method can be found in references [11] and [22]. The model is arranged in an open water tank, and transiently accelerated and reaches the launch speed by the pressure wave. The duration of acceleration process is about 50 μs, which is much shorter than the cavity formation process. Typical photographs are obtained by using a high-speed camera (Phantom® v2512) with a sampling frequency of 5×10^4 frames per second. The time-averaged velocity of the model in the whole process is about 19.1 m/s. The cavitation number is

\[ \sigma = \frac{p_\infty - p_v}{\frac{1}{2} \rho U^2} = 0.54 \]  

2. NUMERICAL METHODS

The numerical method was established with large-eddy simulation (LES) and volume of fraction (VOF) approach.

2.1 Governing equations

Multiphase flow equations are widely used to solve the water-liquid/water-vapor two phases flow problems. The governing equation and momentum equation are,

\[ \frac{\partial \varphi}{\partial t} + \frac{\partial (\rho u_i)}{\partial x_j} = 0 \]  

\[ \frac{\partial (\rho u_i)}{\partial t} + \frac{\partial (\rho u_i u_j)}{\partial x_j} = - \frac{\partial p}{\partial x_j} + \frac{\partial}{\partial x_j} \left( \frac{\mu}{\rho} \frac{\partial u_i}{\partial x_j} \right) \]  

where \( u_i \) is velocity component in \( i \) direction, \( \rho \) is the mixture density, \( p \) is pressure, \( \mu \) is the laminar viscosity.

The mixture density \( \rho \) is defined as,

\[ \rho = (1 - \alpha_v) \rho_l + \alpha_v \rho_v \]  

And \( \mu \) can be defined as,

\[ \mu = (1 - \alpha_v) \mu_l + \alpha_v \mu_v \]  

where \( \alpha \) is the volume fraction of the different phases, \( l \) and \( v \) represent liquid water and vapor.

The transport equation of the water vapor volume fraction is,

\[ \frac{\partial (\alpha_v \rho_v)}{\partial t} + \frac{\partial (\alpha_v \rho_v u_i)}{\partial x_j} = m^+ - m^- \]  

The mass transfer rates of evaporation and condensation \( m^+ \) and \( m^- \) are derived from the Reyleigh-Plesset bubble dynamics equations by Zwart et al. [23].

\[ m^+ = F_{vap} \frac{3 a_{nuc} (1 - \alpha_v) \rho_v}{R_B} \left( \frac{2 \max(p_v - p, 0)}{\rho_l} \right) \]  

\[ m^- = F_{cond} \frac{3 a_v \rho_v}{R_B} \left( \frac{2 \max(p - p_v, 0)}{\rho_l} \right) \]  

where \( R_B = 10^{-4} m \) is the generalised bubble radius, \( p_v = 2340 Pa \) is the saturated vapor pressure, \( a_{nuc} = 5 \times 10^{-4} \) is the nucleation site volume fraction, \( F_{vap} = 50 \) is the evaporation coefficient, \( F_{cond} = 0.01 \) is the condensation coefficient.

2.2 LES approach
LES equations are derived from the above equations (2) and (3) by applying a Favre-filtering operation,
\[
\frac{\partial \rho}{\partial t} + \frac{\partial (\rho \vec{u}_j)}{\partial x_j} = 0
\]

\[
\frac{\partial (\rho \vec{u}_j)}{\partial t} + \frac{\partial (\rho \vec{u}_j \vec{u}_i)}{\partial x_j} = - \frac{\partial \rho}{\partial x_j} + \frac{\partial}{\partial x_j} \left( \mu \frac{\partial \vec{u}_i}{\partial x_j} \right) - \frac{\partial \tau_{ij}}{\partial x_j} \tag{9}
\]

where \(\tau_{ij}\) is the subgrid scale (SGS) stress, which is defined as,
\[
\tau_{ij} = \rho (\vec{u}_i \vec{u}_j - \bar{\vec{u}}_i \bar{\vec{u}}_j) \tag{10}
\]

Based on the Boussinesq equation, the SGS can be computed from,
\[
\tau_{ij} = - \frac{1}{2} \tau_{kk} \delta_{ij} = -2 \mu_t S_{ij} \tag{11}
\]

where \(\mu_t\) is the SGS turbulent viscosity, \(\tau_{kk}\) is the isotropic part, \(S_{ij}\) is the rate-of-strain which is defined as,
\[
S_{ij} = \frac{1}{2} \left( \frac{\partial \vec{u}_i}{\partial x_j} + \frac{\partial \vec{u}_j}{\partial x_i} \right) \tag{12}
\]

In the WALE model the SGS turbulent viscosity is modeled by,
\[
\mu_t = \rho \Delta_s \left( \frac{S_{ij}^d S_{ij}^d}{S_{ij} S_{ij}^d} \right)^{3/2} \tag{13}
\]

\[
\Delta_s = C_w V^{1/3} \left( \frac{S_{ij}^d S_{ij}^d}{S_{ij}^d} \right)^{5/4}
\]

where \(\Delta_s = C_w V^{1/3} \left( \frac{S_{ij}^d S_{ij}^d}{S_{ij}^d} \right)^{5/4}\)

\[
\delta_{ij} = \frac{\partial \bar{u}_i}{\partial x_j}
\]

The constant \(C_w = 0.325\).

### 2.3 Mesh and boundary conditions

A semi-infinite model is adopted by considering that the effect of the tail on the head of the model is neglectable. The inlet and outlet boundaries are placed far enough from the model, and the velocity and pressure boundary conditions are adopted to them, respectively (as shown in Fig.3). The time-averaged velocity 19.1 m/s in the experiment is set as the inflow velocity in the numerical simulation.

![Fig.3 Computational domain and Cartesian mesh in the symmetry cutting-plane](image)

A Cartesian cut-cell mesh is generated by using commercial software ANSYS Meshing, which easily achieves local refinement and perfect orthogonality and is suitable for complex geometries[24, 25]. The mesh in a cutting plane is as shown in Fig.4. 15 layers of inflation are generated around the model. The height of the first layer is set as 1×10^{-4} m with a growth rate of 1.1. Fig.4 shows that the spatial volume grids composing of cubes and inflation layers near the model. The overall cell number is about 450 million. Similar grid resolutions were used in simulations of cavitating flows around circle cylinders with grid independence analysis [26].

![Fig.4 Mesh in the symmetry cutting-plane near the head of test model](image)

### 2.4 Numerical procedures

The LES approach and the wall-adapting local-eddy viscosity (WALE) model are adopted to simulate the turbulent flow. The other detailed numerical schemes and parameters are shown in the following table 1. The commercial CFD solver Fluent is used to perform the simulation.

<table>
<thead>
<tr>
<th>Scheme in time</th>
<th>Second-order implicit</th>
</tr>
</thead>
<tbody>
<tr>
<td>Pressure interpolation scheme</td>
<td>PRESTO!</td>
</tr>
<tr>
<td>Scheme in momentum</td>
<td>Bounded Central Differencing</td>
</tr>
<tr>
<td>Scheme in volume fraction</td>
<td>Modified HRK</td>
</tr>
<tr>
<td>Time step size</td>
<td>1×10^{-5}s</td>
</tr>
</tbody>
</table>

### 3. RESULTS AND DISCUSSION

Similar to the conventional sheet/cloud cavitating flow around axisymmetric bodies, the whole evolution process can be also divided into several stages including growth, re-entry jet, shedding and so on[11, 12]. In the present condition, the cavity shape is closely related to the distribution of ridges and channels, and several particular phenomena occur. We will analyze the characteristic of the cavity development and discuss the mechanisms in different stages.

#### 3.1 Cavity growth stage

Fig.5 shows the cavity shape in the growth stage in experimental and numerical results. Initially, transparent cavity incets and grows. The trailing edge of the cavity shapes like a wave, in which the parts around ridges are shorter, while the parts around channels are longer (as shown in Fig.5, t=1ms & t=2ms). When the cavity length is similar with the model diameter, instable foam-like features show near the cavity closure in the channel in experimental results. At the same time, it can be also observed that liquid water flows from trailing edge to the leading edge of the cavity near the model wall as a re-entry jet in the numerical results (as shown in Fig.5, t=3ms).

![Fig.3 Computational domain and Cartesian mesh in the symmetry cutting-plane](image)
Fig. 5 Cavity evolutions in the growth stage in experimental and numerical results. The iso-surface in numerical results represents where the volume fraction of liquid water is 0.9.

The approximately stable shape at the moment of 3ms is chosen to investigate the cause of cavity formation. So, several slices normal the axial direction are made as shown in Fig.6. The shape of the cavity in the leading edge is the same as the model section, and gradually becomes thicker at downstream. The liquid water around the cavity flows from the ridge to the channel shown by the tangential velocity field in the section (1) in Fig.3. In the following downstream sections, the direction of tangential velocity field changes. The liquid water outside the cavity flows from channels to ridges, while vapor inside the cavity flows from ridges to channels. Consequently, the cavity profile turns into a circular shape in the downstream sections (3) & (4). Then the cavity becomes thinner and disappears near the ridges in the following sections (5) & (6). The cavity profiles in these sections also show petals shapes but the ridges of the cavity profile are near the channel of the model. At the cavity closure, vortexes are generated in the sections, which is closely related to the following U-type local shedding cavities.

Fig. 6 Cavity profiles and tangential velocity in slices normal to the axial direction (x coordinate). Yellow lines represent where volume fractions of liquid phase are 0.1, 0.5 and 0.9, respectively. Flood color represents tangential velocity normal to x coordinate.

3.2 Re-entry jet and local shedding stage

Cavity developments in the next stage are shown in Fig.7. Re-entry jets which show as foam-like in the experimental results continue to flow upstream towards the leading edge. Local shedding phenomena occur repeatedly for several times in both experimental and numerical results. The U-type shedding cavities are entrained to the downstream along the channels as main instable features. These types of cavity structures were demonstrated to be caused by re-entry jets especially the side re-entry jets occurring along the convex region of the attached cavity [27].

Fig. 7 Cavity evolutions in the re-entry jet and local shedding stage in experimental and numerical results. The iso-surface in numerical results represents where the volume fraction of liquid water is 0.9.

The vortex shedding (as shown in Fig.8) is evidently similar to the shedding of cavity. At the moment of 4ms, a half vortex ring is generated which is connected to the main cavity. Then the half ring sheds and deforms into a U-type vortex (as shown in Fig.8 t=5ms & 6ms). A new half vortex ring is observed next to the main cavity closure when the last U-type vortex moves far away (as shown in Fig.8 t=7ms).
Fig. 8 Iso-surfaces where Q-criterion value is $2 \times 10^5$. The flood contour represents the axial velocity (along x coordinate).

And it can be also seen from the Q contour field that the strongest vortex cores appear just around the U-type shedding cavities (as shown in Fig. 9).

Fig. 9 Cavity shapes and Q-criterion values in sections normal to the axial direction. The iso-surface represents where the volume fraction of liquid water is 0.9, and flood contour represents the value of Q-criterion value.

The velocity field demonstrates that the shedding U-type cavity/vortex is generated jointly by the re-entry jet and tangential vortex (as shown in Fig. 10). Because cavity parts near ridges are shorter, the pressure is higher at the cavity closure on the ridges. Consequently, the flow direction is from the ridge to the channel in the section, which induces the tangential vortex in the aforementioned sections.

Fig. 10 Cavity shapes and flow fields at the cavity closure. The iso-surface represents where the volume fraction of liquid water is 0.9, and flood contour represents the value of Q-criterion value, and vectors show the tangential velocity in the section normal to the axial direction.

3.3 Primarily shedding stage
As the re-entry reaches the leading edge of the cavity, the primarily shedding phenomenon occurs. And the shedding cavity shows like an integral ring, which is similar to the conventional cavity shedding process in cloud cavitating flows around axisymmetric cylinders (as shown in Fig. 11). The shedding cavity appears a little difference between places near the ridges and channels. Cavity parts near the ridges are slightly shorter and thinner than the channel parts.

Fig. 11 Cavity evolutions in the primarily shedding stage in experimental and numerical results. The iso-surface in numerical results represents where the volume fraction of liquid water is 0.9.
4. CONCLUSIONS

Some interesting features of the sheet/cloud cavitating flow around a slender cylinder with a petals-shaped section are obtained through experimental and numerical methods. Results agree well between both the methods, and indicate the following conclusions.

In the growth stage, the cavity is longer near the model channel. The cavity profile in sections near the leading and trailing edges normal to the axial direction are various petals shaped, but is circular shaped in the middle section where the cavity is thickest.

In the re-entry jet movement stage, the local U-type cavity shedding occurs repeatedly, which is jointly induced by the re-entry jet and tangential vortex.

The cavity sheds as an integral ring after that the re-entry jet reaches the leading edge of the cavity, which is similar to conventional shedding phenomena in sheet/cloud cavitating flows around slender axisymmetric cylinders.

It can be seen that, for bodies which have grooves or channels along the flowing direction similar with the cylinder in the present paper, cavitation instability may be more remarkable in channels, while the cavity parts near the ridges are more stable.

Moreover, by using dimensional analysis method, governing parameters can be easily conducted including cavitation number and geometry parameters, such as the number of ridges, the ratio of inner diameter to outer diameter. Detailed parametric study should be performed to develop more insight into the flow behavior and further understand the impact of stability in the future.

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