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Thermo-mechanical responses of cracked quasi-transparent film to laser irradiation



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ABSTRACT

Thermal and mechanical responses of a cracked transparent conducting film to laser irradiation were investigated. A representative coin-shaped crack was modeled in the transparent conducting film and the case of vertical incidence of laser beam considered. Firstly, the multiple reflections, transmissions as well as absorptions are formulated in the quasi-multilayer media due to the effect of inner crack. Then, the temperature characteristics generated by the dissipated light energy are computed for the film of typical thermal boundary conditions. Finally, the thermal stress and stress intensity factors around the crack tip are particularly analyzed. The effects of the coin-shaped crack are discussed on the multi-physical responses of the film to laser irradiation.

1. Introduction

The transparent conducting film of Al-doped ZnO (AZO film) have been largely used as in the photovoltaic cells, waveguides and many other important photoelectric devices because of its excellent performance and relatively low price [3,13,10,33,6,23,7,31,9]. Of course, the AZO film is actually not absolute transparent as there is necessarily absorption of light transmitting through it. The electrical conductivity as well as surface morphology of Al-doped ZnO film could be enhanced obviously by increasing its thickness [1]. Although the rough surface of AZO layer could lead to inner discontinuity, which could act as defects especially when the bi-layer structure of AZO/ZnO is taken into concept [17]. These defects are always strictly unacceptable in optical materials as the discontinuity would increase the reflection and scattering of the incident light and dissipate much light energy into heat, which means that unexpected temperature elevation and thermal stress would arise in the structure. Such thermo-mechanical responses would slightly change with the absorption of guided waves if the wave-guide structures are taken into account. Therefore, the correct mathematical descriptions are required on the characteristics of temperature, thermal stress as well as its possible singularity to estimate the reliability of a device involving such AZO film, in particular when tiny flaws appear.

The thermo-mechanical properties of the intact AZO film has been modeled based on the public data distributed in the large literatures [18,8,11 and 21]. However, subsurface or interface discontinuity could be developed in the optical glasses as well as various surface functional films by many processing wherein the contact pressure is involved, or as aforementioned [5,20,22,34]. At the same time, the responses, such as scattering and intensification of the flaws to the incident light have also been utilized to detect them [27,40]. In these detection methods, the mechanical stress effect has also attracting much attention [28]. Although, the full physical mechanism on the interaction between a subsurface crack and light irradiation has not been revealed thoroughly except for some simplified model that approximating the incident light energy as simple thermal loadings without considering the reflection,

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Nomenclature		r_f	upper bound for the nodes selected for curve-fit- ting	
A_0	amplitude of the plane harmonic wave	t	time	
A_k	amplitude of the reflected beams	г Т	temperature	
A_k^t	amplitude of the transmitted beams	T_c	characteristic temperature	
a	model coefficients	$[T_{\rm max}]$	dimensionless maximum temperature	
b	model coefficients	$T_{\rm max}$	maximum temperature	
C_i	model coefficients	Tamb	ambient temperature, 20 °C	
c	specific capacity	u	displacement vector	
D	thermal diffusivity	α	thermal expansion coefficient	
D_i	computing domains	β_i	model coefficients	
d	thickness of the sample	θ_i	angle in medium <i>i</i>	
d_1	crack location, distance from the irradiated surface	λ	laser wavelength	
-	to the crack location	ρ	density	
d_2	distance from the bottom surface to the crack	v	Poisson's ratio	
d_m	center value of d_1 for the averaged calculations	E	surface emissivity, 0.85	
Ε	elastic modulus	σ	Stefan–Boltzmann constant	
E_i	energy flow	σ_{xy}	<i>xy</i> -component of the stress tensor	
E_I	effective energy density of the incident laser	σ_{yy}	yy -component of the stress tensor	
h	convective heat transfer coefficient, 10 W/m ² /K	σ_c	characteristic stress	
K_c	characteristic Stress intensity factor	$[\sigma_{ij}]$	dimensionless ij -component of the stress tensor	
K_I	stress intensity factor of type I	-		
$[K_I]$	dimensionless Stress intensity factor of type I	Definitio	Definitions of the dimensionless parameters	
K_{II}	stress intensity factor of type II			
$[K_{II}]$	dimensionless Stress intensity factor of type II	Q_c	$\frac{E_I}{d}$	
k_c	thermal conductivity	$[Q_i]$	$ \begin{array}{c} \underline{E_I} \\ d \\ \underline{Q_i} \\ Q_c \\ \underline{T-T_{amb}} \end{array} $	
п	outward normal unit vector of the surface	[T]	Q_c $T - T_{amb}$	
n _i	complex refraction index of the medium <i>i</i>		$T_c - T_{amb}$ $T_{max} - T_{amb}$	
Q	heat generation rate	$[T_{\max}]$	$\frac{T_{max} - T_{amb}}{T_c - T_{amb}}$	
Q_c	characteristic heat generation rate	σ_{c}	$\alpha E(T_c - T_{amb})$	
Q_i	heat generation rate per unit volume for domain	$[\sigma_{ij}]$	$\frac{\sigma_{ij}}{\sigma_c}$	
$[Q_i]$	dimensionless heat generation rate per unit vo-	K_{c}	$\sigma_c \sqrt{R_b \alpha (T_c - T_{amb})}$	
	lume for domain	$[K_I]$	$\frac{K_I}{K_c}$	
R	radius of the sample	$[K_{II}]$	K _C K _{II}	
R_c	radius of the coin-shape crack	[v]]	$\frac{K_{II}}{K_c}$	
R_b	radius of the laser beam			

transmission of the light at the cracks [37].

Laser induced damage in such quasi-transparent optical materials, or commonly known as glass materials as well as the interaction between laser and the flaws in these materials play significant role in optical engineering, especially when the multidisciplinary design optimization is considered wherein light, thermal and mechanical reliability are targeted simultaneously [24,25,32,35].

Actually, the interaction between monochromatic light and opaque material is largely discussed in the field of laser manufacturing [19,12], wherein the temperature, phase change and stress are particularly focused on [29,26]. For materials with a

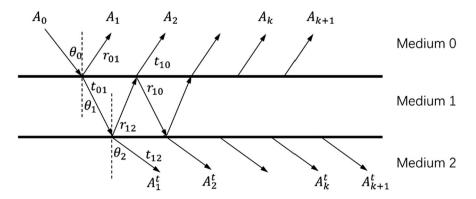


Fig. 1. Sketch of monochromatic light propagation through homogeneous medium.

relatively low extinction coefficient, such as glass and quasi-transparent metal oxide, the penetration depth of laser beam in such materials is much larger than opaque one. Modeling laser irradiation as a surface flux would deviate the mathematical model from the actual physical process and detailed description on the volumetric heat generation is required. Actually, the volumetric heat generation resulted from laser irradiation has been investigated extensively in the intact medium [4,2,38,36].

However, in the presence of defects such as a crack in the medium [15,16,39], energy absorption would be obviously altered because of the reflections at the interface of the medium and the defects. The effect of the defects on volumetric heat source should be carefully inspected to obtain an accurate solution.

In the present work, a single layer of AZO with a crack in the center subjected to a vertically incident laser beam irradiation is investigated. Reflections and transmissions of the laser, as well as interferences of the laser beams, are taken into account. The volumetric heat generation rate due to laser irradiation is formulated, based on which a thermal field is calculated with finite element method. The effects of such thermal field on the induced stress field in the presence of the crack are discussed.

2. Theoretical formula on volumetric heat generation rate

Fig. 1 illustrates the modeling concept developed by Wu et al. [36,38] on the propagation of the monochromatic light in a multilayer media. Many derivations that included in the above reference are rearranged herein to improve the readability of the present work.

A plane simple harmonic electromagnetic wave, with an amplitude A_0 , impinges at the planar interface separating the incident medium (M0) and a second medium (M1) of finite thickness *d*. Without loss of generality, the third medium (M2) underneath the second medium is assumed generic. The complex refraction index for each medium M*i*, *i* = 0, 1, 2 is n_i -*j* k_i and the incident angle in M0 is θ_0 . Herein, we need establish the overall description on the reflection, absorption and transmission of the incident monochromatic light through the medium M2.

The vector summation over all of the spatially overlapped rays is required to calculate the amplitude of the total reflection and the transmission waves. The amplitudes of the reflection rays and the transmission rays are denoted by A_i and A_i^t respectively. The summation of amplitudes for all reflection rays and all transmission rays could be expressed as follows, respectively [38,36].

$$\frac{A_R}{A_0} = \frac{1}{A_0} \sum_{k=1}^{\infty} A_k = r_{01} + t_{01} r_{12} t_{10} e^{-j2\delta_1} e^{-2\xi_1} + \dots + t_{01} (r_{12} r_{10})^{k-2} r_{12} t_{10} e^{-j2(k-1)\delta_1} e^{-2(k-1)\xi_1} + \dots = \frac{r_{01} + r_{12} e^{-j2\delta_1} e^{-2\xi_1}}{1 + r_{01} r_{12} e^{-j2\delta_1} e^{-2\xi_1}}$$
(1)

$$\frac{A_T}{A_0} = \frac{1}{A_0} \sum_{k=1}^{\infty} A_k^t = t_{01} t_{12} e^{-j\delta_1} e^{-\xi_1} + \dots + t_{01} (r_{12} r_{10})^{k-1} t_{12} e^{-j(2k-1)\delta_1} e^{-(2k-1)\xi_1} + \dots = \frac{t_{01} t_{12} e^{-j\delta_1} e^{-\xi_1}}{1 + r_{01} r_{12} e^{-j\delta_1} e^{-2\xi_1}}$$
(2)

Where,

$$\delta_1 = \frac{2\pi}{\lambda} n_1 d_1 \cos \theta_1 \tag{3}$$

$$\xi_1 = \frac{2\pi\kappa_1}{\lambda} n_1 d_1 \tag{4}$$

With r_{01} being the reflected ratio in amplitudes at interface M0-M1 from M0 side and r_{10} M1 side, t_{01} the transmitted ratio across interface from M0 to M1 and t_{10} from M1 to M0.

By introducing the optical admittance η , the reflected and transmitted ratios can be written as

$$\begin{cases} r_{ij} = \frac{\eta_i - \eta_j}{\eta_i + \eta_j} \\ t_{ij} = \frac{2\eta_i}{\eta_i + \eta_j} \zeta_{ij} \end{cases}$$
(5)

where,

N

$$\eta = \begin{cases} \frac{N}{\cos\theta} &, \text{ for } P-\text{ wave} \\ N\cos\theta &, \text{ for } S-\text{wave} \end{cases}$$
(6)

and

$$\zeta_{ij} = \begin{cases} \frac{\cos\theta_i}{\cos\theta_j} &, \text{ for } P-\text{ wave} \\ 1 &, \text{ for } S-\text{wave} \end{cases}$$
(7)

Then after, the energy flux for reflection and transmission could be formulated. Denoting R and T as the reflection ratio and the transmission ratio on energy flow, one can finally get [38,36]

$$R = \left|\frac{A_R}{A_0}\right|^2 = \frac{r_{01}^2 + r_{12}^2 e^{-4\xi_1} + 2r_{01}r_{12}e^{-2\xi_1}\cos 2\delta_1}{1 + (r_{01}r_{12})^2 e^{-4\xi_1} + 2r_{01}r_{12}e^{-2\xi_1}\cos 2\delta_1}$$
(8)

$$T = \left| \frac{A_T}{A_0} \right|^2 \frac{n_2 \cos\theta_2}{n_0 \cos\theta_0} = \frac{(t_{01}t_{12})^2 e^{-2\xi_1}}{1 + (r_{01}r_{12})^2 e^{-4\xi_1} + 2r_{01}r_{12}e^{-2\xi_1} \cos 2\delta_1} \frac{n_2 \cos\theta_2}{n_0 \cos\theta_0}$$
(9)

Since both r_{ij} and t_{ij} differ for P-wave and S-wave, the energy flux ratios should be calculated separately for P-wave and S-wave. Then, the total energy flux ratios can be obtained by adding up the two parts with the following scalar operations.

$$R = \frac{1}{2}(R_P + R_S)$$
(10)
$$T = \frac{1}{2}(T_P + T_S)$$
(11)

where the subscription P and S denote the components for P-wave and S-wave, respectively. Therefore, the heat generation rate *Q* can be obtained through the equation of energy balance as following by assuming that all of the absorbed light energy dissipates into heat immediately.

$$QV = E_i(1 - R - T) \tag{12}$$

where in V represents the volume of the heated region and E_I the energy flow of the incident monochromatic light.

Fig. 2 illustrates the model on the cracked AZO film for the case when a planar crack exists in the medium, dividing the medium into layer 1 and layer 2. Herein, medium 0 and medium 3 refer to the air environment. We can assume the space enclosed by the crack surfaces as another virtual layer of medium denoted by medium c, in particular when the opening displacement of the coinshape crack is very small. This is the very case considered in the present work. Accurately enough, the thickness of the medium c is assumed to be zero and, let R_{c2c} represent the reflection originated in medium c on the interface c-2, and then back into medium c, T_{c23} the transmission from medium c, through medium 2 (layer 2), and into medium 3 (substrate). It is conceivable that the absorption in the crack layer would be zero and optical interference does not occur in such a virtual layer.

For Layer 1, the absorbed light energy rate E_1 can be expressed as

$$\frac{E_1}{E_0} = (1 - R_{010} - T_{01c}) + \sum_{k=2}^{\infty} T_{01c} (R_{c2c} R_{c1c})^{k-2} R_{c2c} (1 - R_{c1c} - T_{c10})$$
(13)

Then the net heat generation per unit time per unit volume in layer 1 is

$$Q_1 = \frac{E_1}{V_1} = \left(\left(1 - R_{010} - T_{01c} \right) + \frac{T_{01c} R_{c2c} \left(1 - R_{c1c} - T_{c10} \right)}{1 - R_{c2c} R_{c1c}} \right) \frac{E_0}{V_1}$$
(14)

where V_1 is the volume of layer 1. Similarly, the absorbed light energy rate E_2 for layer 2 can be obtained as

$$\frac{E_2}{E_0} = \sum_{k=1}^{\infty} T_{01c} (R_{c2c} R_{c1c})^{k-1} (1 - R_{c2c} - T_{c23})$$
(15)

The net heat generation per unit time per unit volume in layer 2 is

$$Q_2 = \frac{E_2}{V_2} = \frac{T_{01c}(1 - R_{c2c} - T_{c23})}{1 - R_{c2c}R_{c1c}} \frac{E_0}{V_2}$$
(16)

where V_2 is the thickness of layer 2.

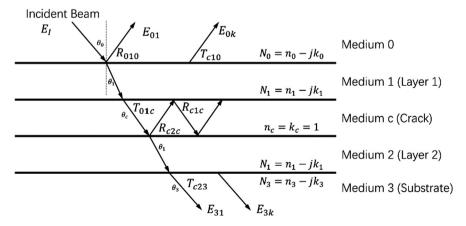


Fig. 2. Illustration of monochromatic light propagation through cracked medium.

3. FEM model on temperature and stress

To calculate the temperature field and stress intensity factors of the coin-shape crack in an AZO film irradiated by a laser beam, an axisymmetric FEM model as shown in Fig. 3 is set up to solve the heat conduction equations as well as the thermo-elastic mechanics equations

$$\frac{1}{D}\frac{\partial T}{\partial t} - \nabla^2 T = \frac{Q}{k_c} \tag{17}$$

And

u

$$\frac{3(1-\nu)}{1+\nu}\nabla\nabla\cdot\boldsymbol{u} - \frac{3(1-2\nu)}{2(1+\nu)}\nabla\times\nabla\times\boldsymbol{u} = \alpha\nabla T$$
(18)

The thermal boundary conditions of

$$k_c \frac{\partial T}{\partial n} + h(T - T_{amb}) = 0 \tag{19}$$

$$k_c \frac{\partial T}{\partial \mathbf{n}} + \epsilon \,\sigma \left(T^4 - T_{amb}^4\right) = 0 \tag{20}$$

are applied to all of the outward surfaces while the adiabatic conditions are assumed on the crack surfaces.

A cylindrical coordinate system is adopted with the origin being coincident with the intersection point of the laser beam centerline and the irradiated surface of the film sample, and z-axis is the symmetry axis with symmetry boundary conditions.

Moreover, the displacement boundary conditions of

$$z = 0$$
 (21)

Was applied to the bottom surface of the model for a rigid substrate.

The laser beam with radius R_b and output energy rate (power) is assumed to be perpendicular to the irradiated surface, which means that $\theta_0 = 0$ in all of the formulae in the previous chapter. The thickness and radius of the film sample is denoted with *d* and *R*, respectively. A coin-shape crack assumed to be located at $z = d_1$ and of half-length R_c .

The overall computational domain can be divided into four adjacent subdomains as following

$$D_0 = \{r, z | R_c \leqslant r < R_b, -d \leqslant z \leqslant 0\}$$

$$\tag{22}$$

$$D_1 = \{r, z | r < R_c, -d_1 \le z \le 0\}$$
(23)

$$D_2 = \{r, z | r < R_c, -d \le z < -d_1\}$$
(24)

and

$$D_3 = \{r, z | r \ge R_b\}.$$
(25)

Apparently, if the crack radius R_c exceeds the laser beam radius R_b , domain D_0 vanishes. Table 1 lists the detailed parameters used for the simulation.

Variations on both crack location d_1 and crack radius R_c and their effects on the thermos-mechanical behaviors of the sample are specifically analyzed. The effects of the various crack location would work at two different scales, in which the long-range effect account for light absorption path length and the short-range for the light interference pattern. To study both the long-range effect and the short-range effect of the crack location on, the crack location d_1 is set to be a function of light wavelength as follows:

$$d_1 = 10\beta_1 + 0.05\beta_2\lambda \tag{26}$$

where β_1 and β_2 are integers successively chosen from 2 to 8 and from -10 to 10, respectively. The radius of the coin-shape crack radius is set as 50 µm, 100 µm, 150 µm, 200 µm, 250 µm, 300 µm, 400 µm, 500 µm, 600 µm, 1000 µm and 1500 µm in succession. Therefore, total 1617 calculation cases are performed in the present work to reveal the latent parameter dependency of the thermo-

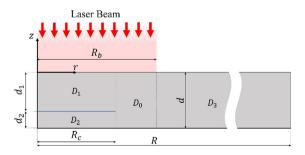


Fig. 3. Sketch of FEM model on thermal and mechanical responses.

Parameter	Value
Laser wavelength λ	523 nm
Effective power density E_I	$1e6 W/m^2$
Radius of laser beam R _b	250 µm
Refraction index of environment (medium 0) n_0	1 - 0j
Refraction index of sample n_1	1.8699–0.0014486j
Refraction index of crack n _c	1 - 0j
Refraction index of substrate n_s	1 - 0j
Calculation domain $R \times d$	$3000 \mu m \times 100 \mu m$
Conductivity k_c	4.4 W/mK
Specific capacity c	480 J/kgK
Density ρ	5670 kg/m^3
Thermal expansion α	7.2×10^{-6}
Elastic modulus E	110 GPa
Poisson ratio ν	0.3

Table 1Parameters used in the FEM model.

mechanical responses.

4. Results and discussion

4.1. Temperature

Fig. 4 plots the three volumetric heat generation rates: Q_0 , Q_1 , and Q_2 , for subdomain D_0 , D_1 , and D_2 respectively.

Obviously, Q_0 is independent of d_1 as only the case of vertical irradiation is considered herein. Due to the interferences between the incident and reflect rays, the volumetric heat generation rates Q_1 , and Q_2 would oscillate and the mid-value decrease with increasing of crack location d_1 . Several magnified insets in Fig. 4 show exactly the oscillatory patterns of both Q_1 and Q_2 around each typical crack location within a range of $\pm \lambda$. The gradual reduction in mean value of the volumetric heat generation rates with increasing of d_1 should be partially resulted from the increasing of absorption distance in the layer above the crack. The two limiting cases that $d_1 = 0$ and $d_1 = 100 \,\mu\text{m}$ are in corresponding to the physical reality that there is no crack, for which Q_1 and Q_2 should be identical to Q_0 , as shown in Fig. 4.

The maximum temperature resides at the top layer of the crack on the axisymmetric axis, as shown in the contour graph in Fig. 5. Fig. 6 shows the maximum temperature in the computational domain versus various crack locations and crack radii, wherein $d_m = 20 \ \mu m$ is the center value of investigated crack locations as a general example. For any given crack location and crack radius, the maximum temperature oscillates in a manner similar to that of volumetric heat generation rate. In the neighborhood of a crack location $d_1 = d_m$, the oscillatory behavior of the maximum temperature can be approximated by the following sinusoidal function:

$$T_{\max} = C_0 + C_1 \sin(ax + b)$$

(27)

where C_0 represents the mid-value, C_1 the oscillation amplitude, *a* the frequency and *b* the starting phase.

The legend in Fig. 6 lists the fitted function for various crack radii, in which the oscillation wavelength is about $\lambda_0/4$, much less than the total depth (*d*) of the sample.

Fig. 7 shows the mid-value of maximum temperature, C_0 as a function of R_c/R_b , with error bars (C_1) indicating the oscillation amplitude of T_{max} . It is indicated that C_0 and C_1 increase quickly with R_c/R_b when $R_c/R_b \leq 1$ and then apparently slowly when $R_c/R_b > 1$, which means that the edge influence would be reduced if the crack tip is far away from the laser beam spot.

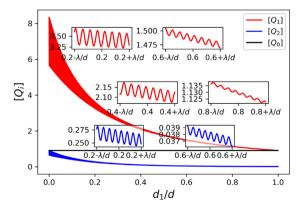


Fig. 4. Fluctuations of dimensionless volumetric heat generation rates $([Q_0], [Q_1], [Q_2])$ versus d_1/d .

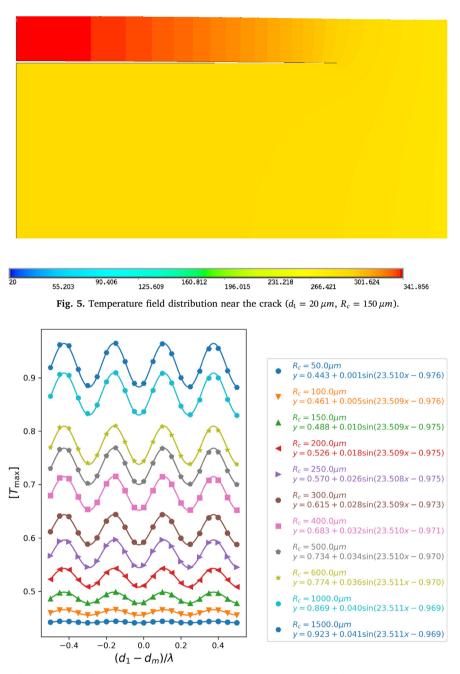


Fig. 6. Dimensionless maximum temperature as a function of d_1 in the neighborhood of $d_m = 20 \, \mu m$.

Fig. 8 shows the dependency of the mid-value of maximum temperature T_{max} on the crack location d_1 for various crack radii. The error bar indicates the rapid oscillation of T_{max} as d_1 changes slightly around each typical point. For any specific crack radius, T_{max} would oscillate rapidly and the mid-value of T_{max} decreases with increase of d_1 . As demonstrated by the data curves in Fig. 4, a larger d_1 means smaller values of Q_1 and Q_2 , i.e. smaller volumetric heat generation rates. Thus for any given crack radius R_c , the maximum temperature T_{max} will decrease as d_1 increases. At the same time, for any given crack location d_1 , a larger crack radius R_c results in a higher maximum temperature. This is because the multiple reflections arise in a larger domain, which indicates more light energy would be absorbed and dissipated into heat.

4.2. Stress

Figs. 9 and 10 show the computed stress contours for the case that $d = 20 \,\mu m$ and $R_c = 150 \,\mu m$.

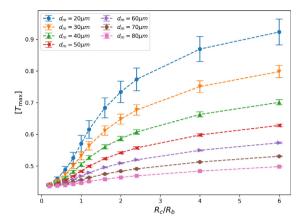


Fig. 7. Dimensionless maximum temperature versus R_c/R_b for various crack location.

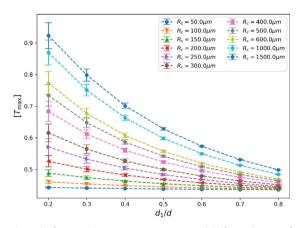


Fig. 8. Dimensionless Maximum temperature versus d_1/d for various crack radius.

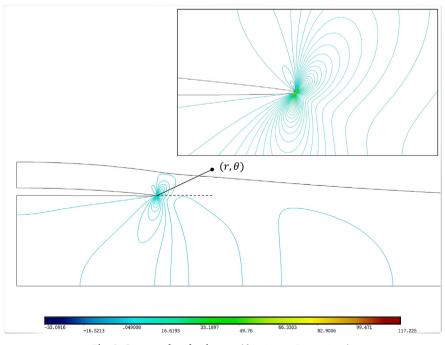


Fig. 9. Contour of σ_{yy} for the case $(d_1 = 20 \ \mu m, R_c = 150 \ \mu m)$.

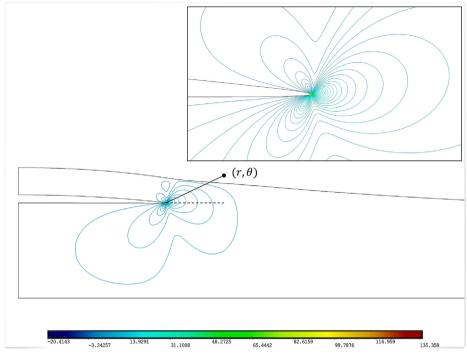


Fig. 10. Contour of σ_{xy} for the case $(d_1 = 20 \ \mu m, R_c = 150 \ \mu m)$.

A polar coordinate system is established with the origin located at the crack tip and the horizontal dash line coincident with $\theta = 0$. It is shown that both stress components σ_{yy} and σ_{xy} are highly concentrated in a very small area of the crack tip, which is generally thought to be stress singularity behavior. Mathematically, the abrupt change of the stress components σ_{yy} and σ_{xy} versus the distance from the crack tip are determined by the stress intensity factors as following according to classic fracture mechanics.

$$\sigma_{yy} = \frac{K_I}{\sqrt{2\pi r}} \tag{28}$$

$$\sigma_{xy} = \frac{\kappa_{II}}{\sqrt{2\pi r}} \tag{29}$$

where K_I and K_{II} are mode I and mode II stress intensity factors, respectively. Therefore, the stress intensity factors K_I and K_{II} could be calculated by fitting the stress data along any path pointing to the crack tip, for which the horizontal path mapping of the stress components for the present computation case are further shown in Fig. 11 as an example. Based on such numerical methods, the stress intensity factors K_I and K_{II} are calculated for the typical cases and the parameter dependency extensively studied as the stress intensity factors dominate the defects behaviors.

Figs. 12 and 13 show the effects of the crack location d_1 on the stress intensity factors K_I and K_{II} , respectively.

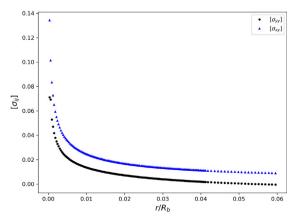


Fig. 11. Dimensionless stresses versus r when $\theta = 0$ for the case $(d = 20 \,\mu m, R_c = 150 \,\mu m)$.

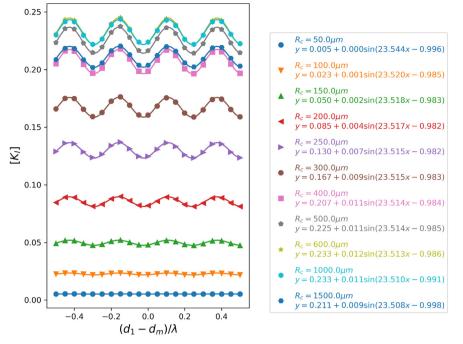


Fig. 12. Dimensionless stress intensity factor $[K_I]$ as a function of d_1 in the neighborhood of $d_m = 20 \, \mu m$.

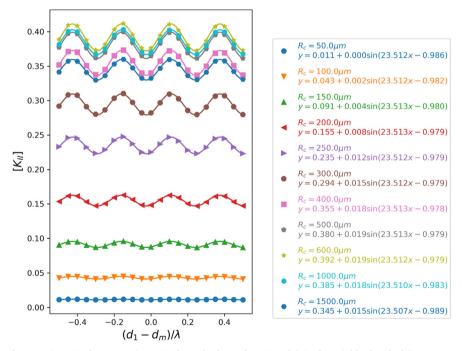


Fig. 13. Dimensionless Stress intensity factor $[K_{II}]$ as a function of d_1 in the neighborhood of $d_m = 20 \, \mu m$.

Sinusoidal oscillation patterns similar to that of temperature peaks can be found again for both K_I and K_{II} in the neighborhood of $d_m = 20 \,\mu m$, for every given crack radius. Therefore, the general sinusoidal function Eq. (27) can also be utilized for data fitting and the fitted expressions are obtained as shown in the right legends in Figs. 12 and 13. It is implied that the slight fluctuations of crack location would result in obvious change of stress intensity factors, which determine the stability of the defects embedded in the film. As we know, the crack would propagate further were the stress intensity factor greater than the fracture toughness of the material. Moreover, it is shown that the magnitudes of both K_I and K_{II} are comparable to the fracture toughness of AZO, which indicates that the fracture would be of mixed mode.

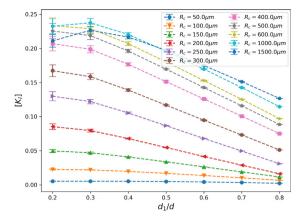


Fig. 14. Dimensionless Stress intensity factor $[K_I]$ as a function of d_1/d in the overall range.

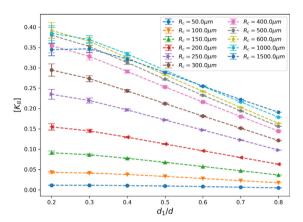


Fig. 15. Dimensionless Stress intensity factor $[K_{II}]$ as a function of d_1/d in the overall range.

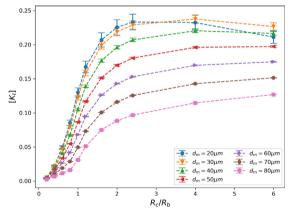


Fig. 16. Dimensionless Stress intensity factor $[K_I]$ as a function of R_c/R_b .

Figs. 14 and 15 show the fluctuations of K_I and K_{II} as d_1 changes from 10 µm to 90 µm, in which the mid-value and oscillation amplitudes are adopted considering the strong oscillation characteristic of both K_I and K_{II} in the vicinity of any given crack location d_1 .

The dashed lines show the relations of K_I and K_{II} with d_1 for various crack radius and the error bars the oscillation amplitudes. It is shown that the oscillation amplitudes of both K_I and K_{II} decay with increasing in d_1 for any given crack radius R_c . It is also shown that on the whole, the intensity factors decrease with increasing of crack depth d_1 , although a small inflection point appears around the location $d_1/d = 0.3$. This could be due to the fact that the temperature gradient has its maximum therein, although the mid-value of maximum temperature is monotonically decrease with increasing of d_1/d as shown in Fig. 8.

Figs. 16 and 17 show the mid-value of the stress intensity factors K_I and K_{II} as functions of the crack radius. Again, the error bars

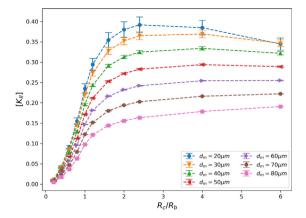


Fig. 17. Dimensionless Stress intensity factor $[K_{II}]$ as a function of R_c/R_b .

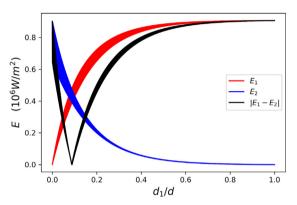


Fig. 18. Energy Fluxes versus d_1/d .

indicates the oscillation amplitude of the stress intensity factors. It is shown that the stress intensity factors increase with the increasing of crack radius R_c in almost all of the present parameter range, which could be divided in two stages according to the climbing rates. The stress intensity factors increase quickly before $R_c/R_b = 1$, then relatively slowly after $R_c/R_b > 1$.

There is also a small deviation from the overall tendency when the crack locates near the irradiated surface, for which the midvalues of K_I and K_{II} drop gradually after $R_c/R_b > 2$ as indicated by the blue $d_m = 20 \,\mu m$ line in Figs. 16 and 17. Stress concentrations in these cases are determined by various amounts of energy absorption of the layers. The energy fluxes as a function of crack position are shown in Fig. 18. When the crack is near the surface $(d_1 < 10 \,\mu m)$, most of the inflow laser energy is absorbed by the layer under the crack. When the crack moves deeper, the dominant absorption domain is then D_1 , the layer above the crack. Roughly speaking, since the thermal stress is induced by the temperature gradient, stress concentration can be related to both the local heat generation and the uneven absorption of inflow energy by the layers above and below the crack. When the crack tip is in the vicinity of the laser spot, the local heat flux is dominated by the local heat generation. On the contrary, when the crack tip is far away from the energy source, the uneven absorption becomes a more dominant factor than the local heat generation to the stress concentration around the crack tip. In that case, the stress intensity factor is to a large extent determined by the absolute difference of absorption, $|E_1-E_2|$, which decreases with an increasing d_1/d to the minimal value around $d_1/d \approx 0.1$, and then increases rapidly to a limiting value in $d_1/d \in [0.2, 0.4]$. Such effect accounts for the different tendency of K_I and K_{II} to d_1/d at small d_1/d and large R_c/R_b than others.

5. Conclusions

A subsurface crack in the transparent conducting film would induce multiple reflection, transmission, interference, absorption and therefore complex heat dissipation from absorbed light energy. The volumetric heat generation rate in the upper part of the film above the crack would harmonically fluctuate, in particular when the crack location is close to the irradiated surface. That is, the fluctuation amplitude of the volumetric heat generation rate would quickly decrease were the crack location farther away from upper surface. Such fluctuation pattern in heat generation results in similar fluctuations of the temperature peaks and stress intensity factors. Averagely speaking, the peak temperatures and stress intensity decrease gradually with the increasing in the distances of the crack location from the upper surface. However, the temperature peaks and stress intensity factor increase with increasing in the crack lengths, both of which implies some ultimate steady magnitudes if the crack length is long enough in comparison with laser beam radius.

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